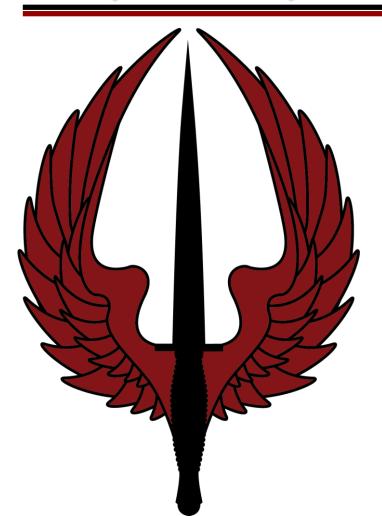
United States Army Special Operations Aviation Command



160th SOAR(A) Troubleshooting Analysis Group

On-wing Wire Diagnostic Practices for Intermittent Symptoms
23 October 2015



 Usually detected by Built-In-Test (BIT) equipment, but also include Crew Observed abnormalities.

• BIT typically monitors functions, not individual wire paths.

 BIT fault thresholds are not published in technical manuals.



- Could Not Duplicate at least once
- Replacement without repeat equals success
- Higher level No Fault Found not reinvestigated
- False removals typically blamed on:
 - Next level testing
 - Technical manual procedures
 - Other end LRU
 - Inaccurate Pilot report
 - Bad wiring



 Most experienced troubleshooters would agree that a degraded Electrical Wiring Interface System (EWIS) could be causal

 How do we determine the serviceability of the EWIS on-wing?



Intermittent Symptom Diagnostics

Not a primary focus during training.

 Fault Isolation Procedures rarely address intermittent symptoms.

 Meters are often ineffective in detecting degraded conditions.

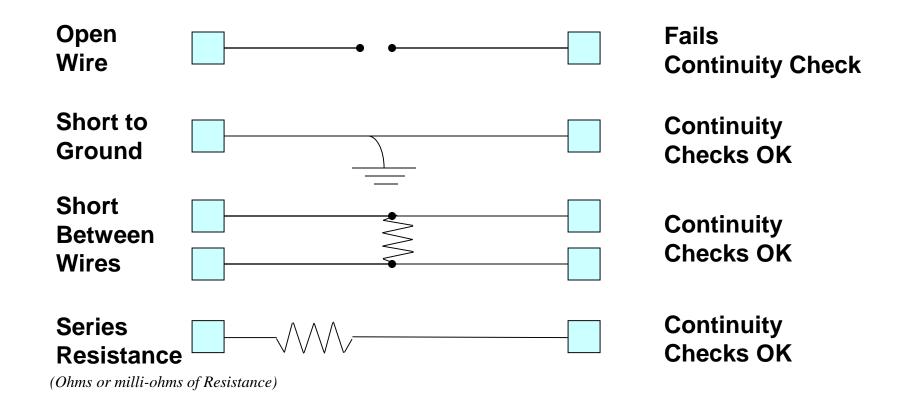


EWIS Degradation

- Loose Connector
- Pin retention
- Cold-cracked solder joints
- Contamination (moisture-fluid-particulates)
- Chaffing
- Corrosion
- Contact degradation
- Grossly over-crimped or under-crimped



Wire Failure Modes





Typical Troubleshooting Steps

- Continuity (<1.0 ohms) using DMM
- Short to ground (OL) using DMM
- Replace sending LRU
- Replace receiving LRU
- Replace Troubleshooter
- Repeat sequence



Other Troubleshooting Steps

- Visual inspection
- Disconnecting, reconnecting
- Cleaning
- Flexing
- Piggyback harness or wire
- Remove and replace



Automatic Wire Test Set (AWTS)

Pros

- Comprehensive
- Flexible
- Repeatable

Cons

- Test Adapter Cables (TACs)
- Test protocol development
- Development process-organization
- Configuration management



Pass/Fail Criteria

Functional

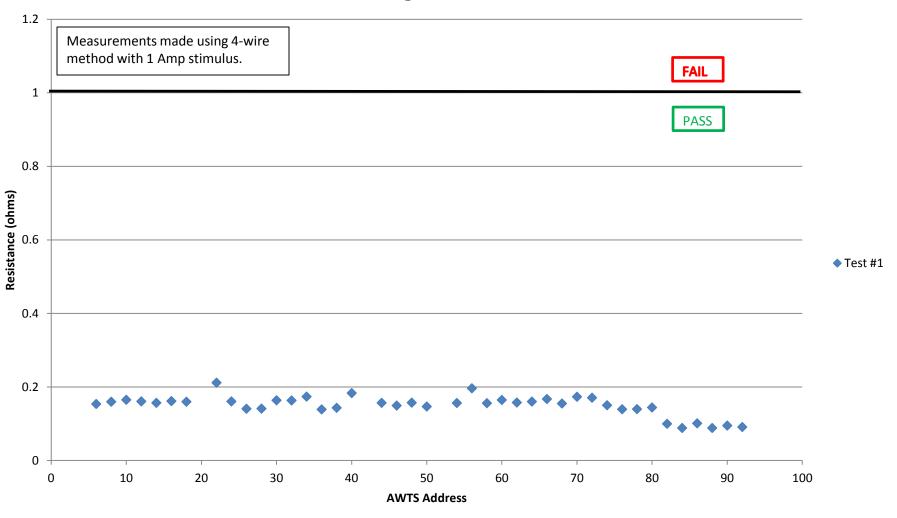
- Established by OEM
- Measured by BIT
- Often proprietary

Material

- Initially established by hardware specification
- Adjusted by recorded measured values
- Diagnostically decisive results

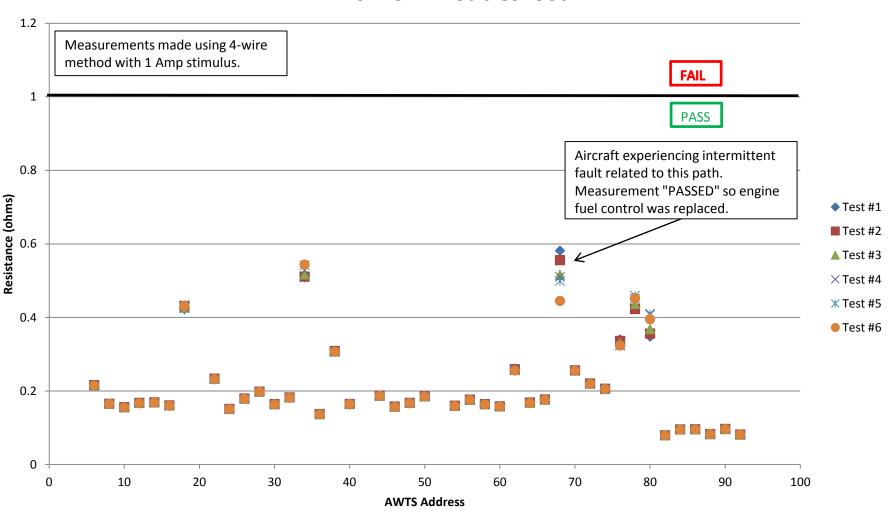


Integration Test



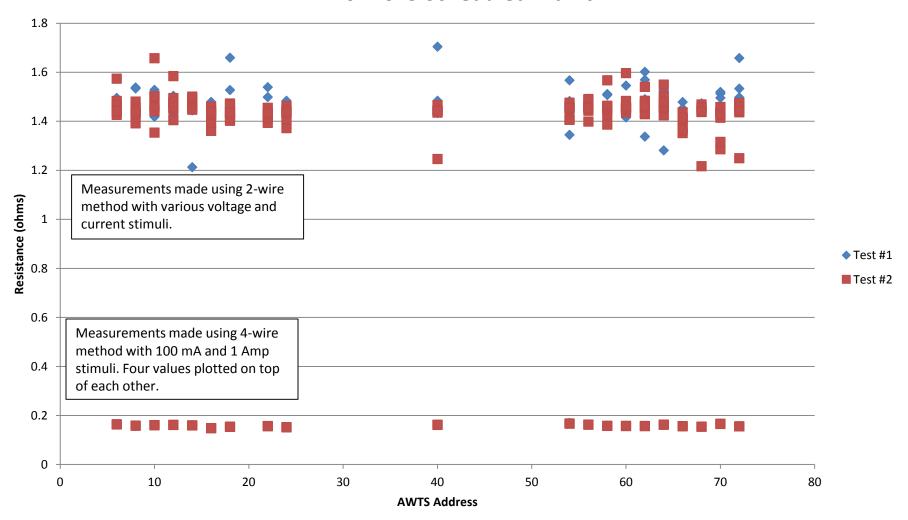


MH-6M 372 Troubleshoot



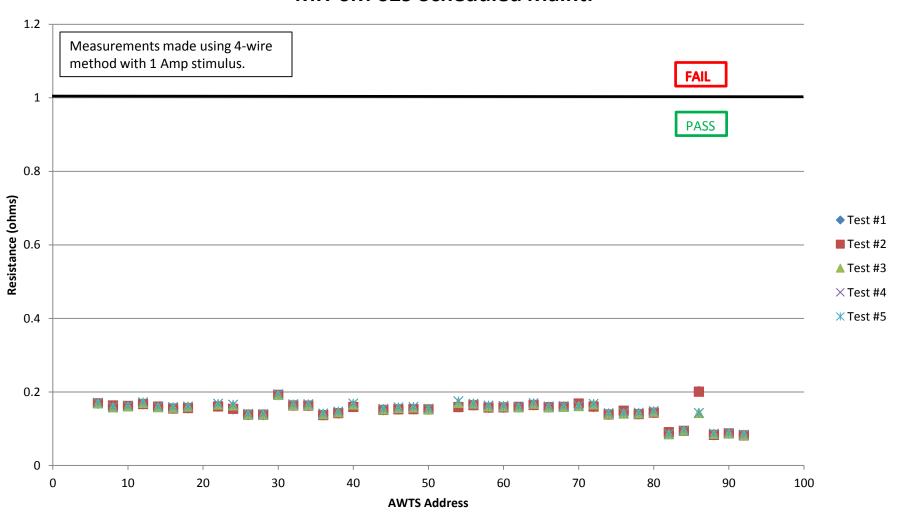


MH-6M 629 Scheduled Maint.



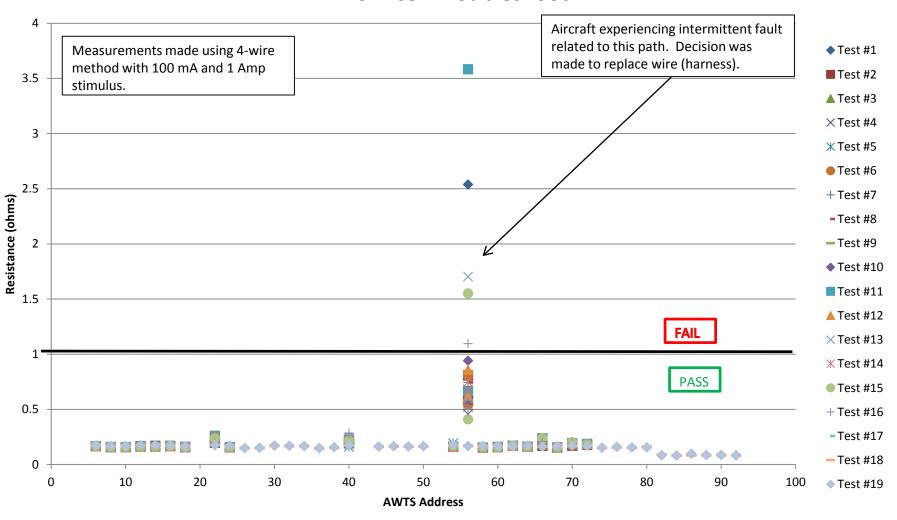


MH-6M 629 Scheduled Maint.



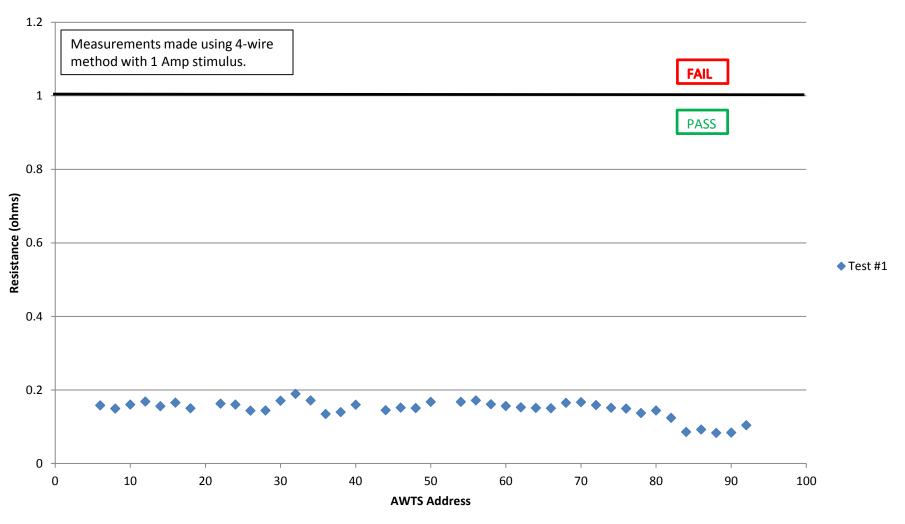


MH-6M 651 Troubleshoot



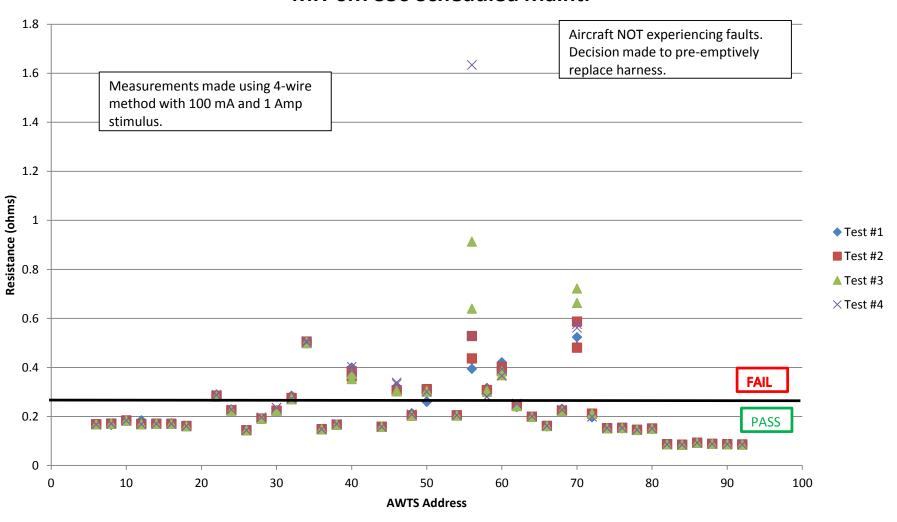


MH-6M 381 Scheduled Maint.



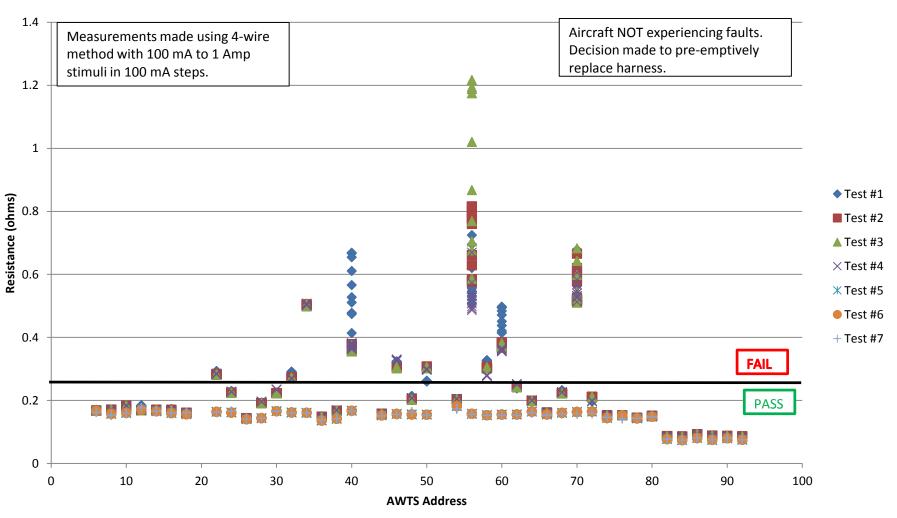


MH-6M 356 Scheduled Maint.



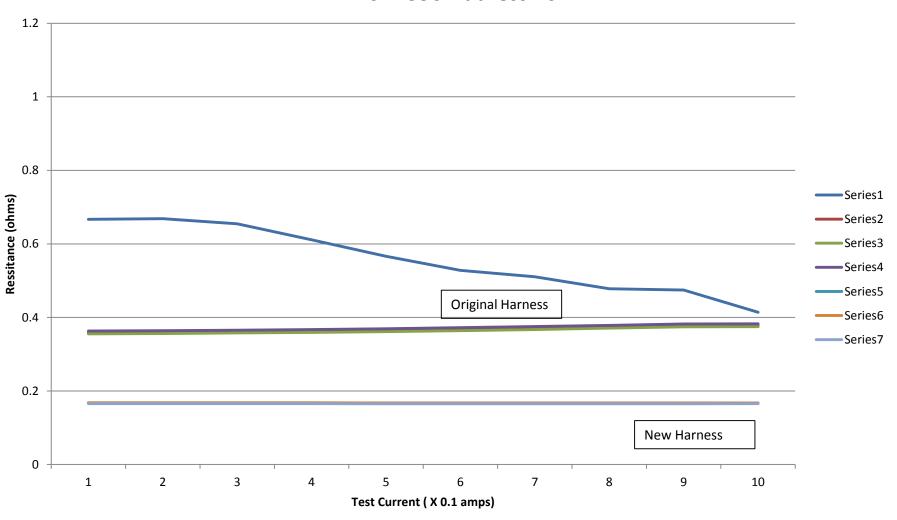


MH-6M 356 Scheduled Maint.





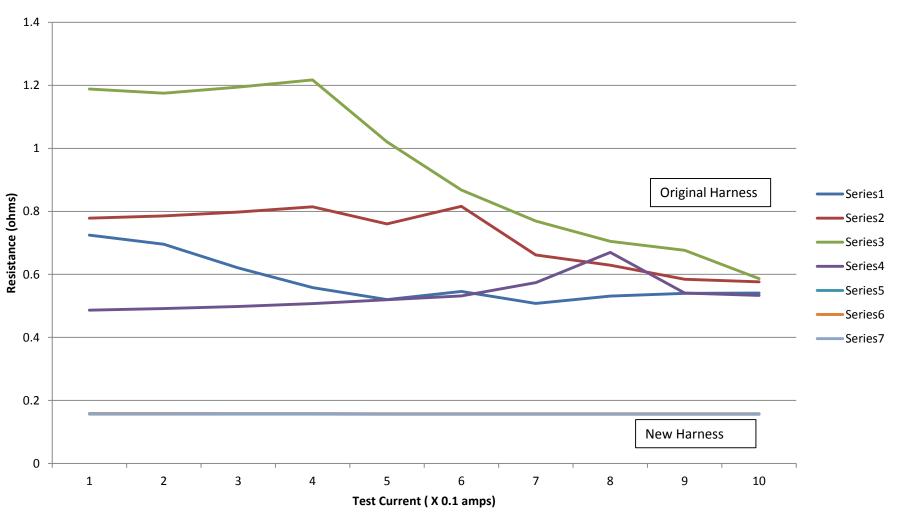






Multi-Stimulus Affects







Fleet Results-Path Forward

- FADEC System EWIS evaluated on 22 aircraft
- 6 Engine Interface Harnesses Replaced since January 2015
- 2 Engine Harnesses Replaced
- 2 Auto/Manual Switches Replaced
- MH-47G FADEC System Test Protocol Set scheduled to be complete by mid November.
- Expect Similar Findings



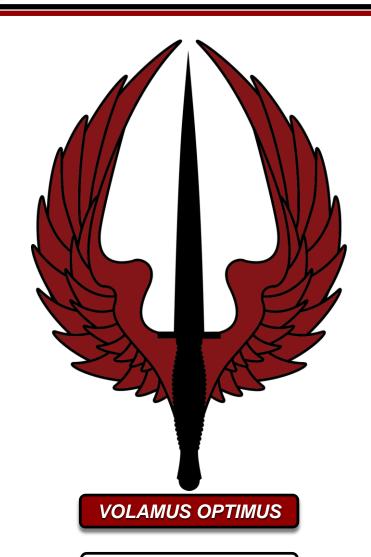
Proposed Certification Protocol

- Low power continuity test: 1 VDC and 5 milli-amps
- High power continuity test: 100 milli-amps and 1 amp
- Low Voltage Insulation Resistance (IR): 10 VDC, 5 milli-amps, flex test
- High Voltage Insulation Resistance (IR): 100VDC, 5 milli-amps; 500 VDC, 5 milli-amps
- All measured values recorded
- Initial condition compared to end state
- If materially uniform, what else needs to be done?



QUESTIONS

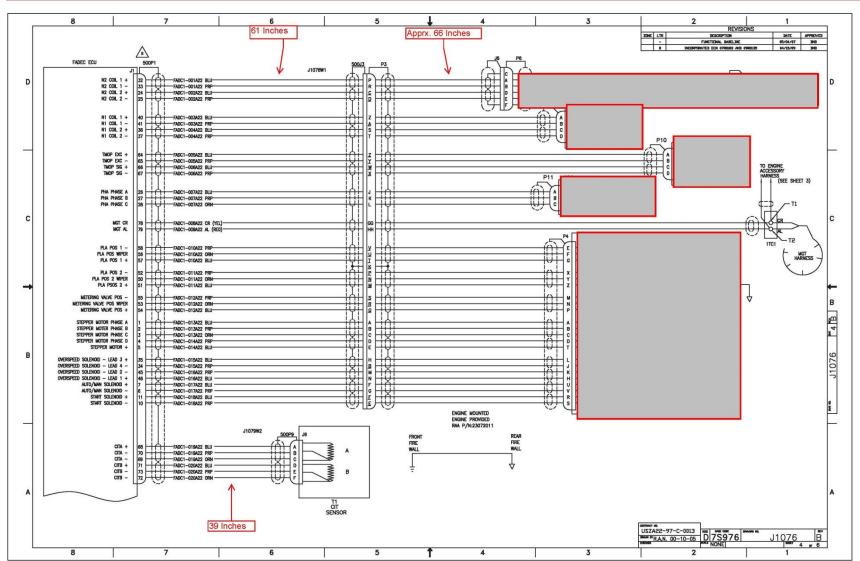
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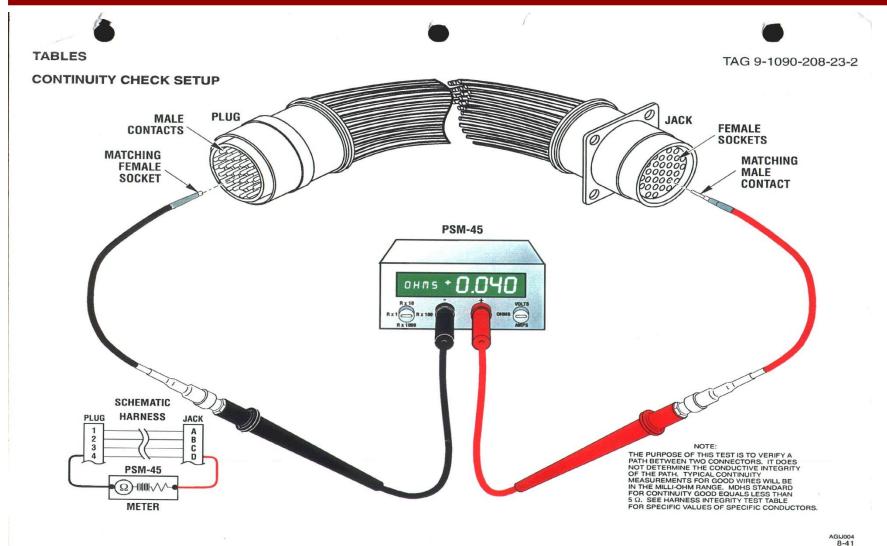
Back Up Slides





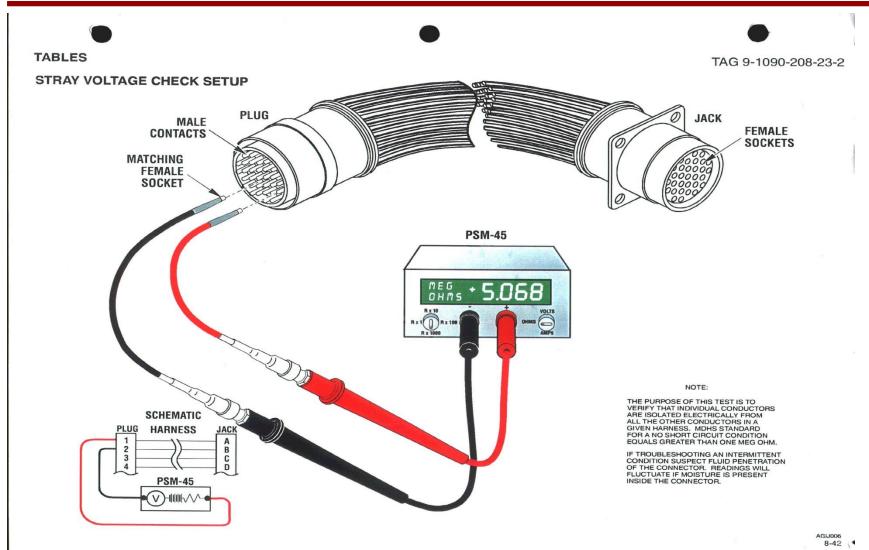


Continuity Test





Short Circuit Test





Voltage Drop Test-Direct

